

# CBCS SCHEME

USN



17CV72

## Seventh Semester B.E. Degree Examination, Feb./Mar. 2022 Design of RC and Steel Structures

Time: 3 hrs.

Max. Marks: 100

- Note:**
1. Answer any TWO full questions choosing one from each module.
  2. Use of IS456, IS800, SP(6), Steel tables are permitted.
  3. Assume any Missing data suitably.

### Module-1

- 1 Design a slab type rectangular combined footing for two columns of size 300mm × 450mm and 300mm × 600mm, subjected to axial loads of 650kN and 900kN respectively. The columns are spaced at 3.6m C/C. The width of the footing is restricted to 1.8m. Use M<sub>20</sub> grade concrete and Fe415 grade steel. Assume SBC of soil  $\Rightarrow 160 \text{ kN/m}^2$ . (50 Marks)

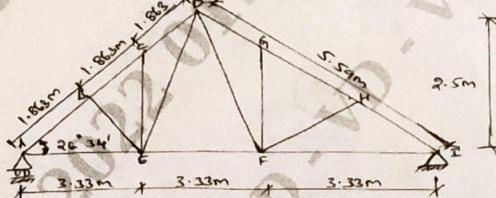
### OR

- 2 An RC portal frame with a hinge base is required to suit the following data :  
 Spacing of portal frames  $\Rightarrow 4\text{m C/C}$  ; Height of column  $\Rightarrow 4\text{m}$   
 Difference between column centers  $\Rightarrow 10\text{m}$  ; Live load on roof  $\Rightarrow 1.5 \text{ kN/m}^2$ .  
 The RC slab is continuous over the portal frame, SBC of soil  $\Rightarrow 200 \text{ kN/m}^2$ .  
 Materials : M<sub>20</sub> and Fe415 steel are used. Design the slab, portal frame and foundation. (50 Marks)

### Module-2

- 3 Design a roof truss shown in Fig. Q3, with forces in each member of the truss is given in table 1. The size of RC column supporting the truss is 300mm × 300mm. Use M20 grade concrete for column. Design the truss using bolt of M16, property class 4.6 for connections and also design anchor bolts. (50 Marks)

Fig.Q3



Member	Design force in kN	
	Compression	Tension
Top chord	54.25	-
Bottom chord	-	48.31
Diagonal (DF, DE)	14.35	-
Member (BE, HF)	-	24.50
Member (CE, GF)	12.40	-

Table - 1.

### OR

- 4 Design a simply supported crane gantry girder for the following data :  
 Span of crane girder  $\Rightarrow 20\text{m}$  ; Span of gantry girder  $\Rightarrow 7\text{m}$  ; Capacity of the crane  $\Rightarrow 220\text{kN}$   
 Self weight of crane excluding the crab  $\Rightarrow 200 \text{ kN}$  ; Weight of crab  $\Rightarrow 60\text{kN}$   
 Wheel base distance  $\Rightarrow 3.4\text{m}$  ; Min hook approach  $\Rightarrow 1.10\text{m}$  ; Self weight of rail  $\Rightarrow 0.3\text{kN/mm}$   
 Height of rail  $\Rightarrow 70\text{mm}$ . (50 Marks)

\* \* \* \* \*

# Scheme of Valuation

sub: Design of RC and Steel structures (17CV72)

Q1.

## Module 1.

Giving Data

column A = 300 x 600 650 kN.

column B, 300 x 600 900 kN

spacing between column = 3.6 m

width of footing = 1.8 m, ABC = 160 kN/m<sup>2</sup>

M20 & Fe 415.

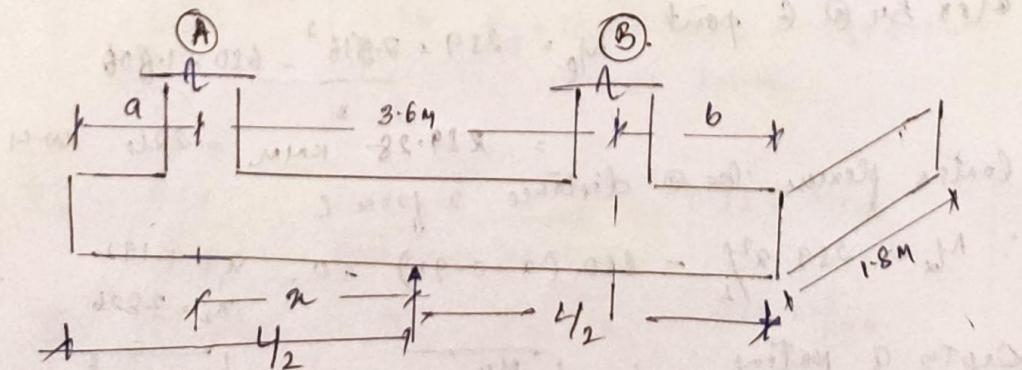
load from Column (A+B) = 1550 kN.

self wt (Cof.) 155

Total load = 1705 kN.

$$\text{Required Area} = \frac{\text{load}}{\text{ABC}} = \frac{1705}{160} = 10.65 \text{ m}^2$$

$$\text{Width of footing} = \frac{\text{Area}}{\text{width}} = \frac{10.65}{1.8} = 5.92 \text{ m} \approx 6 \text{ m}$$



$$\therefore \frac{\text{col B} \times 3.6}{\text{col A} + \text{col B}} = \frac{900 \times 3.6}{900 + 650} = 2.09 \text{ from A.}$$

$$\therefore (a+2) = 4/2 \quad (a+2.09) = 6/2 \quad \& a = 0.91 \text{ m}$$

$$\therefore ((3.6-a)+6) = 4/2 \quad (1.51+b) = 3 \quad \& b = 1.49 \text{ m}$$

$$\text{Uplift pressure} = \frac{\text{Total load}}{\text{Area (pr)}} = \frac{1650}{6 \times 1.8} = 143.33 \text{ kN/m}^2$$

Design of slab for longitudinal bending.

$$\text{as eccentricity } e = 1.84, \quad N = 163.33 \times 1.8 = 289 \text{ kN}$$

Calculating of Shear force & Bending Moment

$$V_{AC} = 289 \times 0.91 = 260.7 \quad V_{AB} = 650 - 289.91 = 360.09 \text{ kN}$$

$$V_{BD} = 289 \times 1.49 = 385.91 \quad V_{BA} = 900 - 385.91 = 514.09 \text{ kN}$$

Shear becomes zero @ E pt.  $289(2) - 600 = 0$

$$\therefore e = \frac{0.91}{289} = 0.0031 \text{ m from footing edge.}$$

$d = 1.800$  from column A centre.

$$M_A = 289 \times 0.91^2 / 2 = 107.2 \text{ kNm.}$$

$$M_B = 289 \times 1.49^2 / 2 = 287.5 \text{ kNm.}$$

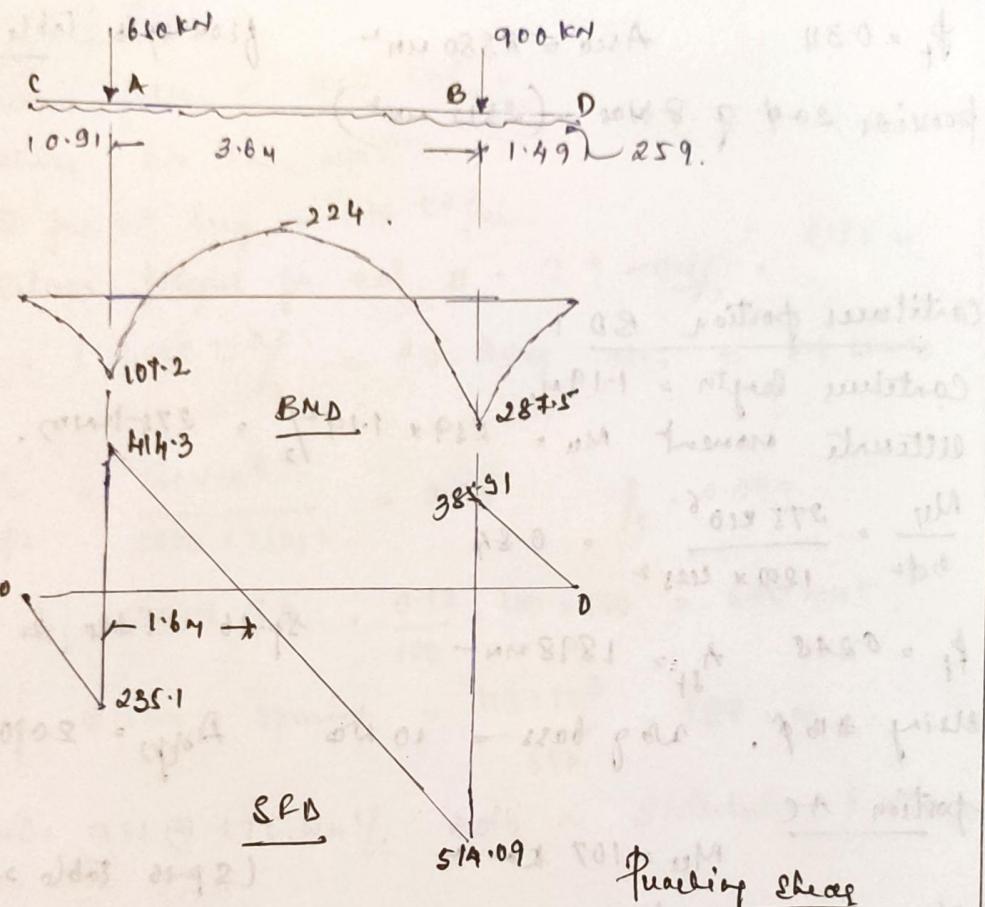
$$\text{Max B.M @ E point.} \quad M_E = 289 \times \frac{0.0031^2}{2} - 650 \times 1.800 \\ = 226.7 \text{ kNm}$$

Central flexure line @ distance  $x$  from C.

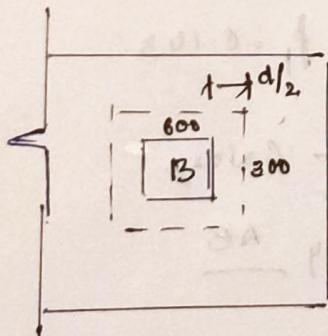
$$\therefore M_x = 289 x^2 / 2 - 650(x - 0.91) = 0. \quad x_1 = 1.192 \quad x_2 = 3.826$$

$$\text{Depth of footing} \quad d = \sqrt{\frac{M_y}{f_u \times b + 0.128}} = \sqrt{\frac{288 \times 10^6 \times 1.8}{8.76 \times 1800}} \\ = 294 + 7.08 = 369 \text{ mm}$$

To have shear take  $D = 500$  &  $d = 425 \text{ mm.}$



Punching Shear



Load on heavy column

$$P_u = 900 \text{ kN} = 1350 \text{ kN}$$

$$P_{us} = 1350 - 143.52 \times 1.5 (0.6 + 0.4) \\ = 1126 \text{ kNm}$$

$$T_r = \frac{P_{us}}{buef}, \quad \frac{1126 \times 10^3}{(725 + 1025) \times 2 \times 425} = 0.757 \text{ MPa}$$

$$T_{ue} = T_{ue} \times k_s \quad T_{ue} = 0.21 \sqrt{20} = 1.11 \text{ N/mm}$$

$$k_s = 0.5 + \frac{340}{600} = 0.9$$

$$T_{ue} = 1.11 \text{ N/mm} > T_r \text{ hence safe}$$

Area of reinforcement:

$$\text{Region AB. } \frac{M_{u4}}{bd^2} = \frac{224 \times 10^6 \times 1.5}{1800 \times 425^2} = 1.03.$$

$$\phi_f = 0.311 \quad \text{Area} = 2380 \text{ mm}^2 \quad \text{from SP16 Table 2}$$

provides 20φ of 8 Nos. (2512 mm<sup>2</sup>)

Concrete portion BD:

$$\text{Concrete length} = 1.19 \text{ m}$$

$$\text{Ultimate moment } M_u = 259 \times 1.19^2 / 2 = 275 \text{ kNm.}$$

$$\frac{M_u}{bdL} = \frac{275 \times 10^6}{1800 \times 425^2} = 0.84$$

$$\phi_f = 0.248 \quad A_{st} = 1898 \text{ mm}^2 \quad \text{SP-16 Table 2}$$

Cleat #16φ. Wog bars = 10 Nos  $A_{stf} = 2010.$

portion AC

$$M_u = 107 \text{ kNm} \quad (\text{SP-16 Table 2})$$

$$\frac{M_u}{bdL} = \frac{160.5 \times 10^6}{1800 \times 425^2} = 0.493. \quad \phi_f = 0.163$$

$$A_{st} = 1096 \text{ mm}^2 \quad \text{provided } 16\phi \text{ of } 6 \text{ Nos.}$$

Check for shear reinforcement between AB

$$V_u = 514 \times 1.5 = 771 \text{ kN.}$$

Shear @ Concrete plazure =  $T_v = 771 - 388 (0.28)$

$$T_v = \frac{V_u}{bq} = \frac{661 \times 10^3}{(1800 \times 425)} = 0.864 < T_{vmax}$$

$$\text{Shear steel } 2512 \text{ mm}^2 = \phi_f = 0.33 \quad \therefore T_e = 0.41$$

Shear reinforcement required.  $\text{SP-16 Table 61.}$

$$\frac{V_{us}}{d} = \frac{(0.864 - 0.41) \times 1800 \times 425}{425} / 1000 = 8.12 / 2 = 4.09$$

provided #12 - 4 legged 180 mm fc.  $\text{SP-16 Table 62.}$

### Design of Slab:

Upward force = 144 kN/m<sup>2</sup>

Concrete b = 1 m, width.

Load per unit bay = 144 kN/m.

Concrete factor for col B = 0.9 - 0.3/2 = 0.75

$$M_u = 144 \times 0.75^2 / 2 = 40 \text{ kNm} \text{ at } 1/4 \text{ = 61 kNm}$$

$$\frac{M_u}{bdL} = \frac{61 \times 10^6}{1000 \times 125^2} = 0.33 \quad f_f = 0.099.$$

$$\text{Min } A_{sf} = 0.12 \times G.A = \frac{0.12}{100} \times 100 \times 500 = 600 \text{ mm}^2.$$

$$\text{Using } \#12 \text{ mm spacing} = \frac{113 \times 10^3}{600} = 188 \text{ mm}$$

Provide #12 @ 175 mm c/c both as distribution & main.

Q2. Given Data:

spacing of frame = 8m      Ht of column = 4m  
 span of frame = 10m      SBC of soil = 200 kN/m  
 live load = 1.5 kN/m<sup>2</sup>

Slab design:

Assume slab depth = 120 mm

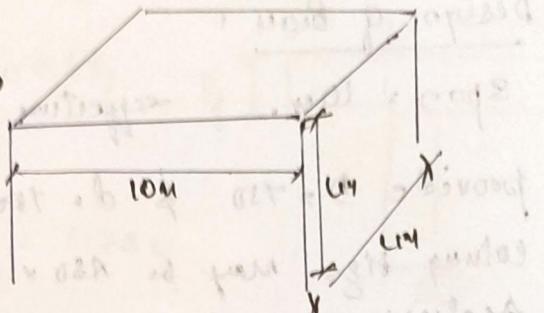
Effective depth = 100 mm

self wt ( $0.12 \times 2.4$ ) = 2.88

pinching wt = 0.75

Total dead wt = 3.63 kN/m

live load = 1.50 kN/m<sup>2</sup>



$$\text{Max } M_u = \frac{W_d L^2}{10} + \frac{W_L \times L^2}{9} = 8.5 \text{ kNm} \quad M_u = 12.75 \text{ kN/m}$$

$$M_{\text{allow}} = Q_{\text{allow}} b \cdot d^2 = \frac{8.78 \times 1000 \times 100^2}{1 \times 10^6} = 27.6 > 12.75$$

$$\frac{M_u}{b d^2} = \frac{12.75 \times 10^6}{1000 \times 100^2} = 1.275 \quad \text{from Sp 16 Table 2}$$

$$P_f = 0.384 \quad f_{st} = \frac{0.384}{100} \times 1000 \times 100 = 384 \text{ N/mm}^2$$

using #10

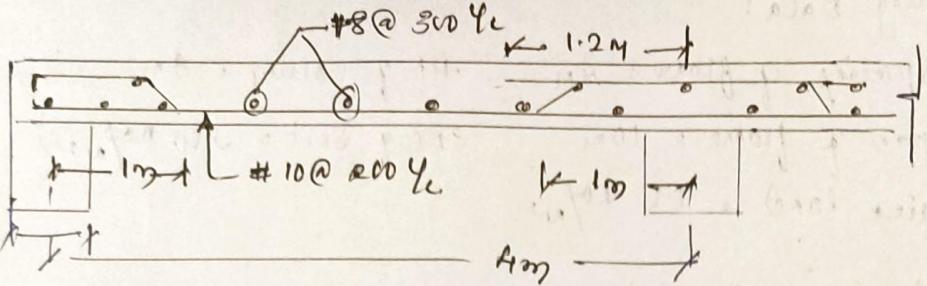
$$\text{Spacing} = \frac{10^4 \times 10^3 \times 1000}{384} = 204$$

provide  $\phi 10 @ 200 \text{ mm}$ .

$$\text{Area of distributing steel} = \frac{0.12 \times 6 \phi}{100} = \frac{0.12 \times 1000 \times 120}{100}$$

Area = 1440 mm<sup>2</sup>

provide  $8 \text{ mm} @ 300 \text{ mm} \text{ clear}$ .



Design of Beam:

$$\text{Span} = 10 \text{ m}, \quad \text{effective depth} = \frac{\text{Span}}{1.3} = 7.69 \approx 7.5 \text{ m}$$

provide  $D = 750$  &  $d = 700 \text{ mm}$ , breath  $b = 150 \text{ mm}$   
column size may be  $180 \times 800 \text{ mm}$ .

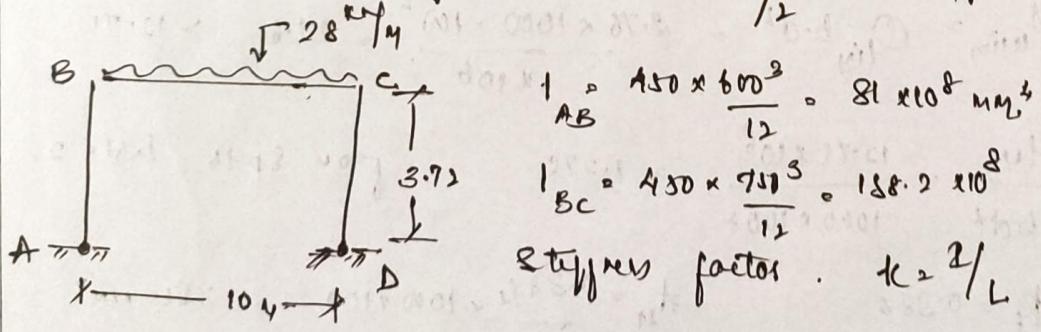
Analysis:

$$\text{load from slab} = (3.63 + 1.5) \times 4 = 20.52 \text{ kN/m}$$

$$\text{self wt of slab} = 0.45 \times 0.63 \times 26 = 6.80$$

$$\text{Total} = \frac{27.32}{2} = 28.0 \text{ kN/m}$$

$$\text{ht of beam above hinge} = 4 + 0.1 - 0.75/2 = 3.72 \text{ m.}$$



$$k_{BA} = \frac{I_{BA}}{L_{BA}} = \frac{81 \times 10^8}{3720} = 21.77 \times 10^5 \text{ N/mm}^2 \quad k_{BC} = \frac{I_{BC}}{L_{BC}} = 15.8 \times 10^5 \text{ N/mm}^2$$

$$\text{Distribution factor } D_{BA} = \frac{k_{AB}}{k_{BA} + k_{BC}} = \frac{k_{BA}}{\sum k_B} = 0.5$$

fixed end moments

$$M_{FAS} = M_{FSR} = M_{FCB} = M_{FOL} = 0.$$

$$M_{FBC} = M_{FCB} = \frac{-28 \times 10^2}{12} = -232 \text{ N-mm}$$

Intermediate Beam is designed using beam becomes critical  
Mid-Span section is designed as T-beam & ends are  
designed as rectangular section.

Mid-Span Beam (T) :  $M_u = 291 \text{ kNm}$

$$\text{flang width } b_f = \frac{b_0}{6} + b_w + 6d_f$$

$$b_0 = 0.7 \times L = 0.7 \times 10 = 7 \text{ m}$$

$$b_f = \frac{7}{6} + 0.45 + 6 \times 0.12 = 2.23 \text{ m}$$

$$\frac{b_f}{b_w} = 5.2$$

$$\frac{D_f}{d} = 0.17$$

from SP16 Table E2.

$T_c = 0.73$ ,

$$\text{Mu}_c = T_c \cdot b_w \cdot d^2 \cdot fck = 0.73 \times 450 \times 700^2 \times 20 = 1896 \text{ kNm}$$

Since  $\text{Mu}_{c,\text{req}} \geq \text{Mu}$  Safe & Single beam

$$\frac{\text{Mu}}{b d^2} = \frac{291 \times 10^6}{(450 \times 700^2)} = 1.8 \quad \text{from SP16 Table 2.}$$

$$\gamma_f = 0.892$$

$$A_{sf} = \frac{0.892 \times 450 \times 700}{100} = 1284 \text{ mm}^2$$

$$\text{Using #20, No of bars} = \frac{1284}{\pi/4 \times 20^2} = 3.93$$

Provide #20 of 4 Nos @ mid-span bottom.

End span:

$$\frac{\text{Mu}_{18}}{b d^2} = \frac{234 \times 10^6}{450 \times 700^2} = 1.1 \quad \gamma_f = 0.827$$

$$A_{sf} = \frac{0.827 \times 450 \times 700}{100} = 1080, \quad \text{using #20.}$$

$$\text{No of bars} = 3.2$$

Provide #20 of 4 Nos @ top in end.

## Moment Distribution Table

Joint	A	B	C	D		
Member	AG	BA	BC	CB	CD	DC
DF	-	0.5	0.5	0.5	0.5	-
FFM	-	-	-28.3	28.3	-	-
Bal		116.5	116.5	-116.5	-116.5	-
CO.			-58.25	X 58.25		
Bal		29.13	29.13	-29.13	-29.13	
CO.			-14.57	14.57		
Bal		7.29	7.29	-7.29	-7.29	
CO			-3.15	3.15		
Bal		1.83	1.83	-1.83	-1.83	
CO.			-0.92	0.92		
Bal		0.46	0.46	-0.46	-0.46	
Total		155.21 <u>(156)</u>	-156.	156	-156.	

## Bending Moment Diagram

$$M_B = 156 \text{ kNm}$$

$$M_{BC} = 1.5 \times 156 = 234 \text{ kNm}$$

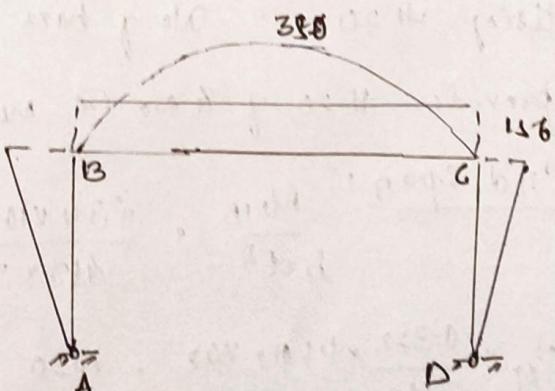
$$\text{Mid-Span Moment} = \frac{w_1 l^3}{8} - 156$$

$$= \frac{28 \times 10^2}{8} - 156 = 194$$

$$M_u = 1.5 \times 194 = 291 \text{ kNm}$$

$$\text{Max Shear @ B or C} = \frac{w_1 l}{2} = \frac{28 \times 10}{2} = 140 \text{ kN.}$$

$$V_u = 1.5 \times 140 = 210 \text{ kN.}$$



(6)

Check for Shear:  $T_v = \frac{V_y}{b d}$ ,  $\frac{210 \times 10^3}{450 \times 700} = 0.67$

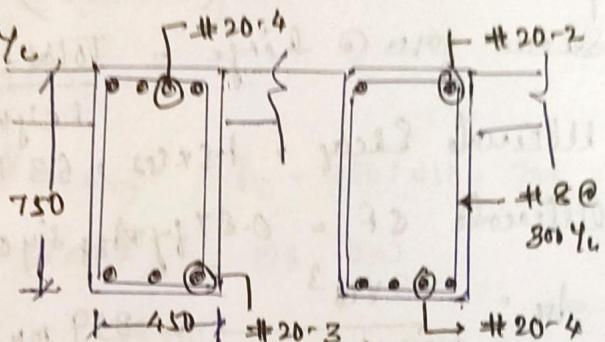
$$f_f = \frac{100 A_{st}}{bd} = \frac{100 \times 1256}{450 \times 700} = 0.39 \approx 0.4 \quad \text{from table 61.}$$

$T_v = 0.632 \times T_u$  provides shear reinforcement.

$$V_{us} = V_u - V_{ue} = 76 \text{ kN}, \quad V_{ue} = \frac{0.632 \times 450 \times 700}{1000} = 186 \text{ kN}$$

$$\therefore \frac{V_{us}}{d} = \frac{76}{70} = 1.0 \quad \rightarrow \text{Table 62 from SP-16.}$$

provide # 8 @ 300 mm  $y_c$ ,



Design of Column:

Ultimate Axial Load =  $P_u = 1.5 \times 140 = 210 \text{ kNm}$ .

Ultimate Moment  $M_u = 1.5 \times 118 = 234 \text{ kNm-m}$

Assume Cover = 50 mm,  $1 \text{ d} \frac{1}{D} = 0.1$

$$\frac{M_u}{\text{fact bD}} = \frac{234 \times 10^6}{20 \times 450 \times 600} = 0.07$$

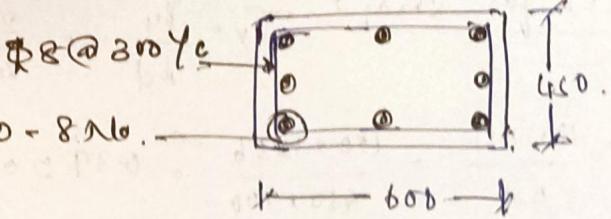
$$\frac{f_y}{\text{fact bD}} = \frac{210 \times 10^3}{20 \times 450 \times 600} = 0.04$$

From chart 22 from SP-16,  $\frac{P}{f_{ck}} = 0.04$ ,  $f = 0.87$ .

$$A_{st} = \frac{0.8 \times 450 \times 600}{100} = 2160 \text{ mm}^2, \text{ No. of bars, } \frac{A}{A_b} = 7$$

provide 8 No. of 20φ.

provide 8 mmφ @ 300  $y_c$  @ ties



### Design of hinge

Permissible comp stress in concrete @ Hinge =  $0.6 \times 0.4 f_{ck}$   
factored thrust = 210 kN.

$$C_s \text{ area of hinge required} = \frac{210 \times 10^3}{16} = 13125 \text{ mm}^2$$

provided,  $200 \times 100$  ( $20000$ ) for hinge.

$$\text{Shear force @ hinge} = \frac{\text{Total moment}}{\text{Length of col}} = \frac{156}{3.72} = 42$$

$$\text{Ultimate shear} = 1.5 \times 42, 63 \text{ kN}$$

$$\text{Ultimate SF} = 0.87 f_y A_{st} s_{iy} \sigma$$

$$A_{st} = 83 \times 10^3$$

$$0.87 \times 415 \times s_{iy} 31^\circ = 839 \text{ MN}^2$$

$$\left. \begin{array}{l} \text{ton} = \frac{80}{50} \\ \& \Phi Q = 31^\circ \end{array} \right.$$

$$\text{provided, } \# 16 - 40 \text{ nos } (\text{A} = 804 \text{ mm}^2)$$

### Design of foundation

$$\text{Axial load} = 140 \text{ kN}$$

$$\text{Column wt} = 24 \text{ kN. } (0.45 \times 0.6 \times 3.72 \times 24)$$

$$\text{Sdfl wt (10\%)} = 16 \text{ kN}$$

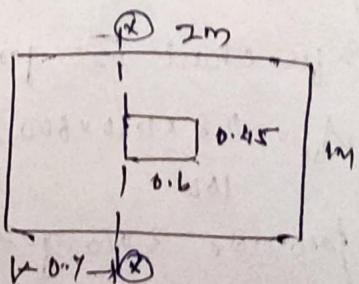
$$\text{Total load} = 180 \text{ kN}$$

$$\text{App area} = \frac{\text{load}}{88c} = \frac{180}{210} = 0.9 \text{ m}^2$$

$$\text{let assume area} = 2 \times 1 \text{ m}$$

$$q_{unr} = \frac{P + M}{A} = 153 \text{ kN/m}^2$$

$$q_{unr} = \frac{P - M}{A} = 87 \text{ kN/m}^2$$



$$\text{Avg pressure } q = \frac{453 + 27}{2} = 90 \text{ kN/m}^2$$

$$BM Q_{\alpha-\alpha} = 90 \times 1 \times 0.7^2 / 2 = 22 \text{ kNm} \quad M_u = 33$$

assume  $D = 300$  &  $d = 250 \text{ mm}$

$$\frac{M_u}{B d L} = \frac{33 \times 10^6}{1000 \times 250^2} = 0.528.$$

$$P_f = 0.15 I. \quad A_{st} = 801 \text{ mm}^2$$

from table 2 sp16

provide #12 @ 300  $\gamma_c$  both ways.

Check for factoring shear:

$$a_0 = 600 + 250 = 850 \text{ mm} \quad b_0 = 450 + 250 = 700.$$

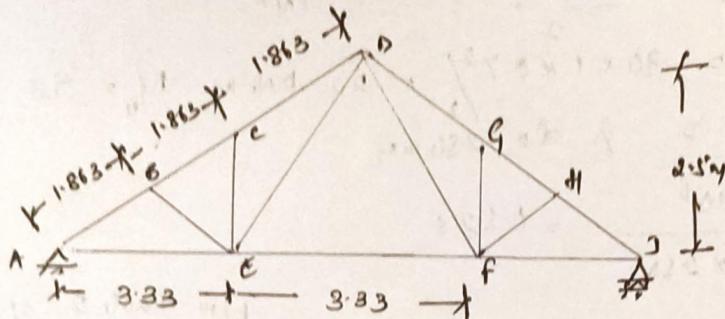
$$V_p = 180 - 90 \times (0.85 \times 0.7) = 126 \text{ kN}.$$

$$T_p = \frac{V_p}{2(a_0+b_0)d} = \frac{126 \times 10^3}{2(850+700)250} = 0.16 \text{ MPa}$$

permissible shear stress  $\sigma_s = 0.25 f_y = 111.8 \text{ MPa} > T_p$ .

Module 2

Q3.



i) Design of top chord AB: compression force  $> 54.25 \text{ kN}$ .

$$\text{length} = 1.863 \text{ m} \quad \text{effective length} = (kL) = 0.7xL = 1.306 \text{ m}$$

Try I.S.A 50x50x6 - 2 Nos placed back to back.

$$I_y = \frac{1306}{15.1} = 86.3 < 180. \quad \text{from Table} \quad A = 1136 \text{ mm}^2$$

$$\text{from IS 800-2007 Table 8cc)} \quad f_y = 250 \text{ N/mm}^2$$

$$X = 0.48.$$

$$f_{cd} = \frac{X f_y}{P_{u0}} = \frac{0.48 \times 250}{1.25} = 112 \text{ N/mm}^2$$

$$\therefore P_d = A \cdot f_{cd} = \frac{1136 \times 112}{1000} = 127 \text{ kN} > 54.25 \text{ kN}$$

ii) Bottom chord AE: Tension force  $= 48.31 \text{ kN}$ .

$$\text{length} = 3.33 \text{ m} \quad \text{effective length} = 0.7xL = 2.33 \text{ m}.$$

try I.S.A 50x50x6 - 2 Nos connected by Gutter pt g flange

$$A = 1136 \text{ mm}^2, S_{min} = 15.1 \text{ mm} \quad \text{Using #16 spaced } 50 \text{ mm } \text{yc.}$$

iii) Design strength due to Yielding  $\Gamma_{dy} = \frac{A_y f_y}{P_{u0}}$

$$\Gamma_{dy} = \frac{1136 \times 250}{1.1} = 258 \text{ kN}.$$

(b) Design strength governed by tearing. Cl. 6.3.3.

$$T_d = \alpha \cdot A_g \frac{f_y}{\gamma_m} \quad A_g = [(50 - 18)(6 \times 2)] = 884 \text{ mm}^2$$

$$\alpha = 0.6.$$

$$T_d = 0.6 \times 884 = \frac{510}{1.25} = 408 \text{ kN} > 48. \text{ Hence safe.}$$

(iii) member DE, compressive load = 14.35 kN.

Effective length 3m. Try 50x30x~~85~~ mm.

$$A = 479 \text{ mm}^2 \quad f_{uy} = 192 \text{ MN} \quad A = \frac{300 \times 0.7}{19.2} = 110.$$

Table 8(c) for IS 800-2007.

$$\phi = 0.291$$

$$f_{ed} = \frac{0.291 \times 250}{1.25} = 58.2.$$

$$P_d = A \cdot f_{ed} = 479 \times 58.2 = 27.8 \text{ kN} > 14.35$$

Hence safe.

(iv) Member DE = Tension force = 24.5 kN.

Try single 50x30x~~85~~ mm. - 6mm thick Guest plate & 2 rows of 16 bolts @ 50 mm c/c.

$$A = 179 \text{ mm}^2. \quad f_{uy} = 152 \text{ MN}$$

$$\text{Using } \#16 \text{ bolts} \quad A_{ue} = (50 - 18) 5 = 160 \text{ mm}^2$$

$$A_{go} = (50 - 8) 5 = 225 \text{ mm}^2$$

$$A_f = 179 \text{ mm}^2$$

(a) Strength governed by Yielding  $T_d = \frac{A_f \cdot f_y}{\gamma_m}$ .

$$T_d = \frac{179 \times 250}{1.1} = 108 \text{ kN.}$$

(b) Strength governed by rupture.

$$T_d = \frac{0.9 A_{ue} f_y}{\gamma_m} + \frac{\beta \cdot A_{go} f_y}{\gamma_m}$$

$$\beta = 0.7$$

$$\beta = 1.4 - 0.176 \left( \frac{w_f}{f_u} \right) \left( \frac{b_f}{L_c} \right)$$

$$q_{dy} = \frac{0.9 \times 100 \times 410}{1.25} + \frac{0.7 \times 225 \times 250}{1.40} = 83.0 \text{ kN}$$

(c) Strength governed by block shear

$$A_{yg} = 5(50+50) - (1.5 \times 18) = 473 \text{ mm}^2$$

$$A_{yg} = 5(50+50) = 500 \text{ mm}^2$$

$$A_{tg} = [(5 \times 25) - (0.5 \times 18)] = 116 \text{ mm}^2$$

$$A_{tg} = 5 \times 25 = 125 \text{ mm}^2$$

$$\begin{aligned} \sigma_{Dg_1} &= \frac{A_{yg} b_f}{13 \cdot Y_{M0}} + \frac{0.9 A_{tg} b_f}{Y_{M1}} \\ &= \left[ \frac{510 \times 250}{13 \times 1.1} + \frac{0.9 \times 116 \times 410}{1.25} \right] = 99.2 \text{ kN} \end{aligned}$$

$$\begin{aligned} \sigma_{Dg_2} &= \frac{0.9 A_{yg} f_u}{13 \cdot Y_{M1}} + \frac{A_{yg} b_f}{Y_{M0}} \\ &= \frac{0.9 \times 473 \times 410}{13 \times 1.25} + \frac{125 \times 250}{1.1} = 109.1 \text{ kN} \end{aligned}$$

Hence all  $\sigma_{Dg} = 83.0$  is lowest  $> 24$  hence safe.

Q4. Given Data : Span of Girder = 204.  
Span of G. Girder = 7m.

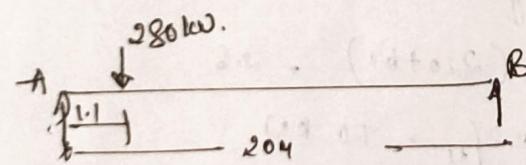
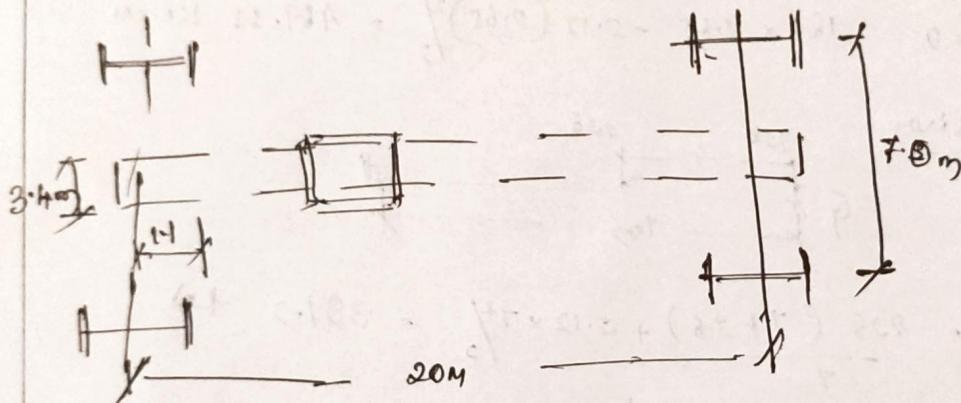
Capacity of Crane = 220 kN.

Self wt of Crane excluding Cab = 200 kN.

wt of Cab = 60 kN. wheel base distance = 3.6 m.

road. approach = 1.104 Self wt of Rail = 0.3 kN/m

ht of Rail = 704 m.



$$R_A + R_B = 280 + 280 = 480 \text{ kN}.$$

$$\therefore R_A = \frac{200}{2} + 280 \left( \frac{18.9}{20} \right) = 364.6 \text{ kN} \quad R_B = 115.4 \text{ kN}.$$

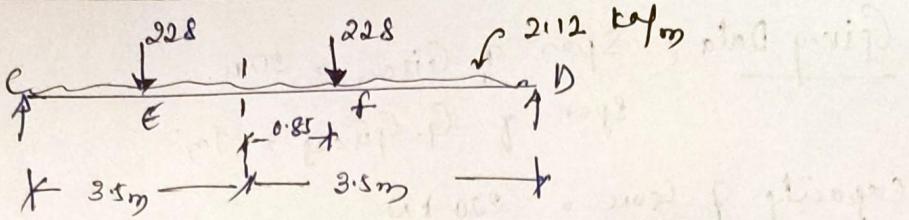
$$\text{load on each wheel} = \frac{364.6}{2} = 182.3 \text{ kN}.$$

$$\text{Add 2% Impact} = 45.62$$

$$\text{Total load} = 228 \text{ kN}.$$

$$\text{self wt of Girder } w_1 = 2 \times \frac{228}{250} = 1.824$$

$$\text{wt of Rail} = 0.3 \text{ kN/m} \quad \text{Total load} = 211.2 \text{ kN}$$



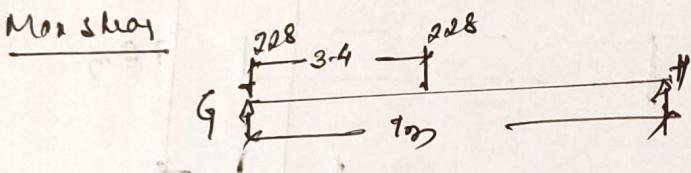
Max BM,  $R_c + R_d = 2 \times 228 + 2.12 \times 7.0 = 470.84 \text{ kN}$

$$R_d = \frac{228 \times 0.95}{7.0} + 228(3.5 + 0.85) + 2.12 \times 7 \frac{1}{2} = 180$$

$$\therefore R_c = 290.80 \text{ kN}$$

$$\Sigma M_f = 0 \quad 180 \times 2.65 - 2.12(2.65) \frac{7}{2} = 469.88 \text{ kNm}$$

Max shear



$$\therefore V_g = \frac{228(7 + 2.6)}{7} + 2.12 \times 7 \frac{1}{2} = 397.2 \text{ kN}$$

Calculator of horizontal force

$$\text{Axial force on rail} = \frac{10}{100}(220 + 60) = 2.8$$

$$\text{load on each wheel} = 2.8 \frac{1}{4} = 0.7 \text{ kN}$$

$$\text{factored load} = 10.5 \text{ kN}$$

$$\therefore M_g = \frac{10.5 \times 469.88}{228} = 21.67 \text{ kNm}$$

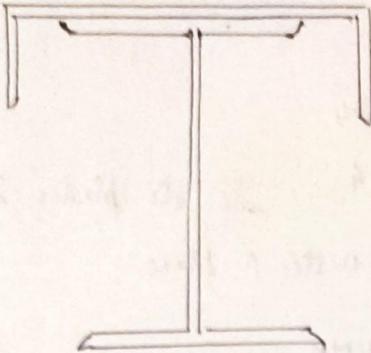
$$\text{Vertical shear due to impact} = 0.25 \times 397 = 99.25$$

$$\text{total shear} = 496.25 \text{ kN}$$

$$\text{preliminary section } d \cdot Y_{12} = \frac{7000}{12} = 583.33 \text{ mm}$$

$$b = Y_{12}^2 \frac{7000}{25} = 28044 \quad \text{provided IWB 600 @ 133.3}$$

$$\& ISMC 300$$



IWB 600	ISMC 300.
170.38	- A - 4360
250	- b - 80.
1061.98.3 x 10 <sup>6</sup>	- I <sub>xx</sub> - 636.26 x 10 <sup>5</sup>
1702.1 x 10 <sup>6</sup>	- I <sub>yy</sub> - 810.8 x 10 <sup>5</sup>
b <sub>y</sub>	- 23.6 mm

$$\gamma = \frac{170.38 \times 310 + 4360 (600 + 7.6 - 23.6)}{170.38 + 4360} = 260 \text{ mm.}$$

$$I_{xx} = 1355.43 \times 10^6 \text{ mm}^4$$

$$Z_{xx} = \frac{I_{xx}}{\gamma} = \frac{1355.43 \times 10^6}{260} = 5185 \times 10^3 \text{ mm}^3.$$

$$\text{Compressive force } @ \gamma_y \quad Q = \frac{bd^3}{12} + b\gamma_p = 9136.0 \times 10^4$$

$$Z_{yy} @ \text{flange} = \frac{I_{yy}}{\gamma} = \frac{9136 \text{ mm}^4}{150} = 609.07 \times 10^3 \text{ mm}^3.$$

Plastic Modulus of Section:

AT @ a distance  $\gamma_p$  from tension flange

$$(\gamma_p - 21.3) \cdot 11.2 + (250 \times 21.3) = \frac{A}{2} \therefore \gamma_p = 510.2 \text{ mm}$$

$M_p$  = Moment of force @ Yielded plastic NA.

$$\therefore M_p = 4686430 \text{ Nm}$$

$$\therefore Z_p = \frac{M_p}{f_y} = 4686430 \text{ mm}^3.$$

for top flange,  $Z_{yy} = \frac{bd^3}{4} + \frac{bd\gamma}{4} \times CQ$ .

$$2 Z_{yy} = 824.96 \text{ mm}^3.$$

### Check for Moment capacity

$$\frac{b}{f} g \text{ IWB } 600 = 5.6 \leq 8.4 \\ -u- = 49.76 \leq 84$$

$\frac{b}{f} g$  channel  $= 6.06 \leq 8.4$ . So its plastic section moment capacity of bending vertical plane.

$$M_{Dz} = f_y \cdot Z_f = 1065.1 \times 10^6 \text{ N-mm}$$

$$M_{Dy} = \frac{1.2 Z_e \cdot f_y}{1.1} = 1026.81 \text{ N-mm} \therefore M_y = \frac{1065}{1.1}$$

top flange:  $M_{Dy} = \frac{f_y \cdot Z_t}{1.1} = 187.4 \text{ kNm}$

$$\frac{1.2 Z_e \cdot f_y}{1.1} = 166.11 \text{ kNm}$$

### Check for local capacity

$$\frac{M_x}{M_{Dz}} + \frac{M_y}{M_{Dy}} \leq 1 \quad \frac{104}{1065} + \frac{21.62}{166.11} = 0.66 + 0.13 = 0.79 \leq 1,$$

### Check for buckling resistance

$$M_D = f_b \cdot Z_p \cdot f_{bd}$$

$$M_D = Z_p \cdot f_{bd} \quad f_{bd} = \frac{1.1 \pi^2 E}{(1/s_y)^2} \left\{ 1 + \frac{1}{20} \left( \frac{L_e/s_y}{L_f/f_y} \right)^2 \right\}^{0.5}$$

$$\therefore L_e \sqrt{1/y_A} = 71.57 \text{ m} \quad f_{bd} = 206 \text{ N/mm}^2$$

from Table 13(a)  $f_{bd}^2 = 159 \text{ N/mm}^2$

$$1 M_{Dz} = 745.4 \text{ kNm} >$$

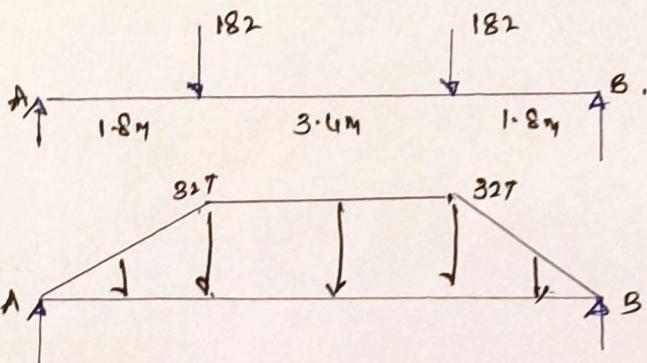
thus safe.

Check for Shear :  $V_x = 190 \text{ kN}$ .

Shear Capacity =  $\frac{A \cdot f_y \cdot w}{S \times 1.1} = 913 > 496$  hence safe.

Check for Deflection :

At working load deflection =  $\gamma_{f_{50}}$ .



Reaction @ Conjugate beam =  $\gamma \cdot \text{total } \frac{M}{EI} \text{ Diagram}$

$$= \left( \frac{\gamma \times 1.8 \times \frac{327}{E_1}}{E_2} \right) + \frac{327}{E_1} \times 1.7 = \frac{851}{E_2}$$

$$E_1 \Delta = 851 \times 1.7 - (0.9 \times 327) \times 2.3 = 327 \times 1.7 \%$$

$$E_1 \Delta = 1829.$$

$$\therefore \Delta = \frac{1829}{10^3} = 1.829 \text{ mm}$$

$$\text{Actual Allowable} = \frac{4}{950} = \frac{7000}{700} = 9.23$$

$\therefore$  Deflection is satisfied.

~~Aff~~ :  
(G.V.C.)

~~Author~~  
07-04-2022